KU-KEN- 760

No.TCD -2819-1-89

Date July 17, 1989

Japan Civil Aviation Bureau

TAIKUSEI-KAIZEN-TSUHO

Airworthiness Directive

The undermentioned examinations or modifications are mandatory.

Fuji Model FA-200 Series Airplanes : except for the 1. Applies to: Bulletin 200-020.

2. Compliance required as indicated, unless already accomplished.

To prevent possible failure of the control column tube in flight due to fatigue cracks at the gust lock pin holes, accomplish the following paragraphs 2.1 thru 2.3.

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  2.1 Within the next 50 flight hours or prior to the next aerobatic flight after the effective date of this AD, whichever occurs first, and thereafter at intervals not to exceed 100 hours time in service, inspect the control column tube gust lock pin hole area for cracks in accordance with paragraph 12 of FHI Service Bulletin No.200-008 Rev.B dated May 30,1989.
- 2.2 If any cracks are found by the inspection required by paragraph 2.1, before further flight replace the control column assembly with a serviceable assembly.
- 2.3 Alternate means of compliance with this AD may be used, if approved by the Director General of Civil Aviation Bureau.
- 3. Remarks
- 3.1 This AD amends AD No.TCD-2819-87 dated August 26,1987.
- This AD becomes effective on August 7,1989.
- 3.3 FHI Service Bulletin No.200-008 Rev.B and FHI Technical Bulletin 200-020, both dated May 30, 1989, pertain to this AD.